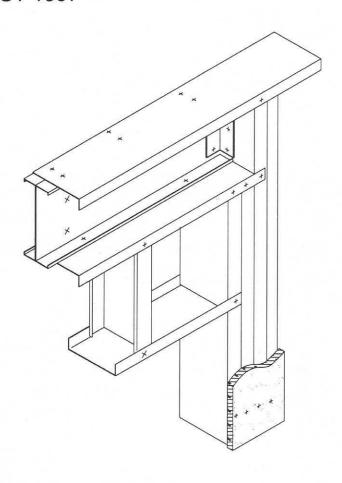


COLD-FORMED STEEL BACK-TO-BACK HEADER ASSEMBLY TESTS

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Steel in Residential Construction



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Prepared for:

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INTRODUCTION

This publication was developed by the National Association of Home Builders Research Center for the American Iron and Steel Institute, the US Department of Housing and Urban Development, and the National Association of Homebuilders. It is intended to provide more affordable design and construction techniques of residential buildings using cold-formed steel framing. AISI believes that the information contained in this publication substantially represents industry practice and related scientific and technical information, but the information is not intended to represent an official position of AISI or to restrict or exclude any other construction or design technique.

American Iron and Steel Institute

The following publication has been developed for the American Iron and Steel Institute (AISI) which is comprised of representatives of steel producers in the US, Canada and Mexico. In the production of this publication, due diligence has been exercised in consulting a wide range of pertinent authorities and experiences and efforts have been made to present accurate, reliable and useful information. AISI acknowledges the principal authors of this publication, Kevin Bielat, Jay Crandell, PE, Nader Elhajj, PE, and Shawn McKee of the National Association of Home Builders, Research Center.

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Cold-Formed Steel Back-to-Back Header Assembly Tests August 1997



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INTRODUCTION

The purpose of this test program was to investigate the structural capacity and performance of built-up headers typically used in cold-formed steel framing. The configuration of the headers tested was limited to doubled, back-to-back C-sections assembled in accordance with the *Prescriptive Method for Residential Cold-Formed Steel Framing* (Prescriptive Method) [1]. Currently, allowable spans of header assemblies are typically determined by doubling the allowable capacity of a single C-section (one of the two member built-up header assembly) as calculated in accordance with the *Specification for the Design of Cold-Formed Steel Structural Members* (AISI Design Specification) [2]. This conservative assumption and simplified design approach under-estimates the actual performance of the header assembly resulting in an uneconomical design and unnecessarily short header spans, particularly with thin deep sections. The findings of this study demonstrate that much greater header spans are possible with improved design rules. Benefits realized from built-up header assemblies are as follows:

- back-to back webs stiffen and support each other against web crippling at concentrated loads;
- back-to-back webs, of thin deep sections, stiffen and support each other against conditions where high shear loads exist; and,
- back-to-back headers produce a doubly-symmetric section which has a coinciding centroid
 and shear center, thus minimizing torsional stability concerns associated with singlysymmetric C-sections.

The major goal of the testing program is to support the development of more economical header designs for cold-formed steel framing. The results are intended to be implemented in future editions of the Prescriptive Method.

LITERATURE REVIEW

A literature review of similar work was performed prior to testing. Little information pertaining to the benefits of back-to-back headers was found in the literature.

N. Hetrakul and W. Yu studied 73 I-beam specimens subjected to a combination of partial edge loading and bending moment. The purpose of the work was to develop an interaction formula to predict the effect of the bending moment on web crippling strength of I-beams (double C-sections) having unreinforced webs [3]. In 1992, Chen and Fang performed an experimental investigation of back-to-back I-beams connected by both resistance spot welding and arc-welding. Their study showed that both the stability and ultimate strength of these beams were different with respect to the two separate welding methods. The beams connected with the arc-welding approach were considerably stronger and more stable [4]. Neither of these reports addressed the objectives of this study.



EXPERIMENTAL APPROACH

A total of 24 back-to-back header assemblies were constructed and tested in accordance with Table 1. Each header consisted of two C-sections fastened back-to-back with two #10 screws spaced 24 inches on center. The top and bottom tracks were fastened to the flanges of the C-sections with two #8 screws spaced at 24 inches on center (one through each flange). A detail of the built-up header assembly is shown in Figure 1. Representative header spans for the tests were selected from the header tables in the Prescriptive Method.

The header test specimens represent common construction practices. All steel used in the tests had a specified minimum tensile strength of 33 ksi. Actual mechanical properties were verified by tensile tests in accordance with ASTM A370-92 [5]. Base steel thicknesses were also determined following ASTM 90-93 [6]. The dimensions of all steel members used in the tests conformed to the Prescriptive Method requirements with the exception of some members falling below the minimum base steel thickness. All mechanical properties and base metal steel thickness were verified from coupon sections cut from the flat portion of the web. The cold-formed steel members were supplied by two manufacturers.

TABLE 1 Header Test

TEST	HEADER CONFIGURATION	MEASURED DIMENSIONS ¹	SPAN (INCHES)	NUMBER OF TESTS
A	2-2x4x33	1 5/8" x 3 ½"	47	3
В	2-2x4x43	1 5/8" x 3 ½"	57	3
C	2-2x8x33	1 5/8" x 8"	36	3 .
D	2-2x8x43	1 5/8" x 8"	80	3
E	2-2x10x43	1 5/8" x 10"	67	3
F	2-2x10x54	1 5/8" x 10"	104	3
G	2-2x12x43	1 5/8" x 12"	57	3
Н	2-2x12x68	1 5/8" x 12"	104	3

Notes:

The header assemblies were tested using a universal testing machine. A heavy steel I-beam and 1.5-inch-wide steel bearing plates were used to apply a two-point concentrated load on the header samples. The ends of the header were supported on 1.5-inch-wide steel plates, representing a minimum typical bearing width (i.e. 1½ inches). The test set-up is illustrated in Figure 2. The load was applied at a load rate of 1/20 inch per minute until the headers failed. Deflections at the midpoint of the header were measured during the full range of loads until failure. The ends of the header were restrained against weak axis rotation and lateral movement as shown in Figure 3. Rotation was allowed in the plane of bending to model a simply supported beam with pinned ends. Rollers were not used at the reactions or concentrated load points because of the intent to replicate conditions in actual use.

All return lips were 1/2-inch. Dimensions were verified within a tolerance of ±1/16 inch.

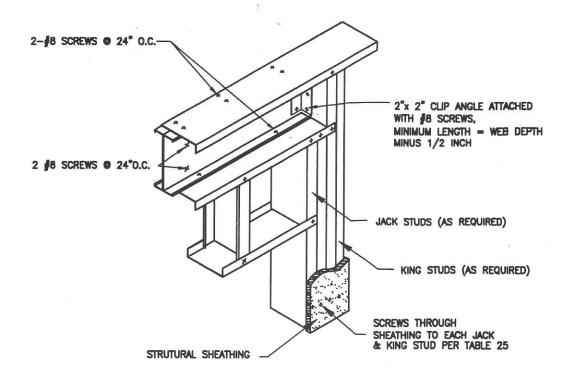


FIGURE 1
Detail of a Built-up Header Assembly

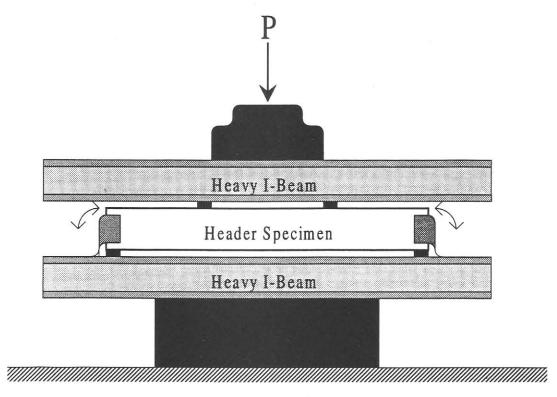


FIGURE 2 Header Test Apparatus

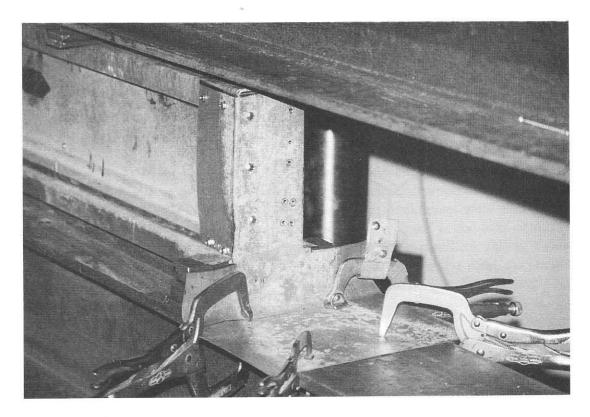


FIGURE 3
End Condition of the Header Specimens

RESULTS

A detailed compilation of the test results can be found in Appendix A. The table in Appendix A reports the following key data from the tests:

- the total load at L/240 deflection (a typical building code deflection limit);
- the ultimate load and standard deviation; and
- the calculated ultimate moment, shear, and total load (i.e. the sum of the 1/3-point loads).

Failure modes, tensile test data and base steel thickness were also reported for each test specimen. In all cases, the failure mode was local buckling of the compression flange initiated at the 1/3-point concentrated loads.

The calculated ultimate moment and calculated ultimate shear were determined in accordance with the *Specification for the Design of Cold-Formed Steel Structural Members* [2]. Section properties were calculated using AISIWIN Version 1.0 cold-formed steel design software [7]. A sample calculation can be found in Appendix B.

Table 2 shows calculated ultimate shear and moment values. These values are based on average yield stresses and base steel thicknesses measured for each of the test specimens (see Appendix A). The composite action of the track members connected to the header was neglected in determining the calculated capacities. The capacity of a single C-section was doubled to estimate the shear and moment capacity of the built-up section. In addition, the top flanges (compression flanges) were considered to be laterally supported at the 1/3-point concentrated loads.

TABLE 2
Calculated moment and Shear values

TEST	HEADER SPECIMEN	SPAN (INCHES)	CALULATED ULTIMATE MOMENT (FT-LBS)	CALULATED ULTIMATE SHEAR (LBS)
A	2-2x4x33	47	1,917	1,343
В	2-2x4x43	57	2,473	2,986
С	2-2x8x33	36	4,964	854
D	2-2x8x43	80	6,772	1,923
Е	2-2x10x43	67	8,552	1607
F	2-2x10x54	104	12,369	3,070
G	2-2x12x43	57	12,972	1,623
Н	2-2x12x68	104	22,776	5,827

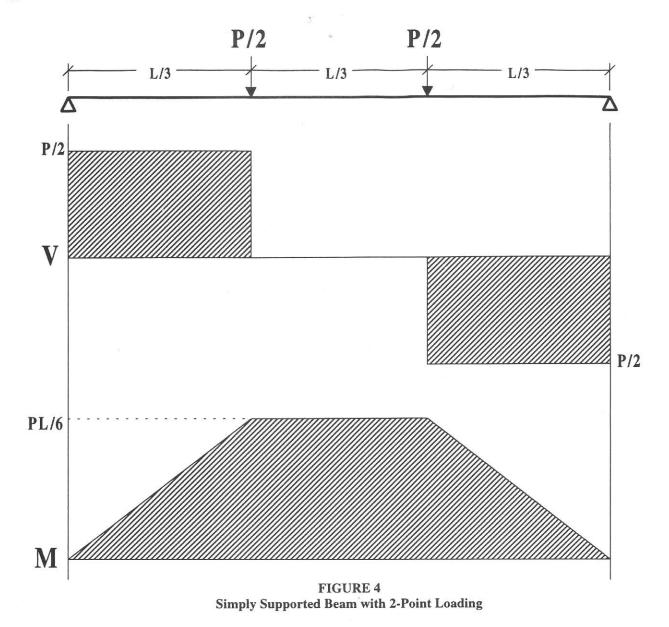
Using the calculated ultimate shear and moments of Table 2, the calculated ultimate load, P_u , may be determined using conventional mechanics of a simply supported beam with two, equal, concentrated loads as shown in Figure 4. The estimated ultimate load produced from the calculated ultimate moment is determined by:

$$P_u = [6*M_u/L] Eq.1$$

 P_u and M_u are the ultimate total load (i.e. the sum of both 1/3-point loads) and ultimate calculated moment, respectively, and L is the clear span length of the header assembly. The estimated ultimate load produced from the maximum shear equation is determined by:

$$P_{u} = [2*V_{u}]$$
 Eq.2

V_u is the ultimate calculated shear.



The test results demonstrate that the calculated values consistently underestimate the tested capacities, particularly for thin deep sections. As shown in the test data in Table 3, the tested ultimate load is greater than the calculated ultimate capacity, with shear controlling the design, by a ratio ranging from 1.41 to 4.30. The following factors contributed to the large ratio of tested ultimate load to calculated ultimate load (shear controlled):

- 1. local bucking of the top (compression) flange at the concentrated load points initiated all failures, not web crippling or overall bending;
- 2. web height to thickness (h/t) ratio effects are limited when webs brace each other (back-to-back members); and
- 3. span length.

TABLE 3
Summary Data and Analysis of Specimens

TEST	HEADER SPECIMEN	SPAN (INCH)	H/T RATIO	TESTED ULTIMATE LOAD¹ (LBS)	CALULATED ULTIMATE LOAD ² (LBS)	LOAD INCREASE RATIO
A	2-2x4x33	47	115	3,899	2,687 shear	1.45
В	2-2x4x43	. 57	86	3.822	3,124 bending	1.22
С	2-2x8x33	36	238	7,341	1,707 shear	4.30
D	2-2x8x43	80	181	6,141	3,846 shear	1.60
Е	2-2x10x43	67	226	9,111	3,214 shear	2.83
F	2-2x10x54	104	183	8,673	6,139 shear	1.41
G	2-2x12x43	57	277	11,723	3,246 shear	3.27
Н	2-2x12x68	104	169	21,413	11,654 shear	1.84

Note:

- Values are based on an average of three tests per specimen.
- Calculated values are based on twice the capacity of a single C-section designed according to the AISI Design Specification. The controlling load effect according to design calculations is noted.

The span length is critical in affecting whether shear or bending controls the design of a header. Shorter spans are typically controlled by shear while longer spans are typically controlled by bending. The fact that the spans were not varied for each member size creates a difficulty in completely analyzing the data without additional testing. As seen in Figure 4, the maximum moment and maximum shear just about coincide at the load points resulting in a potential combined stress problem. Combined bending and shear may have contributed to the failure of the header specimens. Therefore, as the span lengths varied, the relationship between the ratio of predicted bending to predicted shear failure also changed. This partially explains why the specimens with larger h/t ratios and shorter spans had higher load increase ratios and specimens with smaller h/t ratios and longer spans had lower load increase ratios. This trend is demonstrated in the data by comparing the 2-2x8x33 (Test C) and the 2-2x10x54 (Test F) header specimens.

Local buckling of the compression flange was the major visible mode of failure for all specimens. As the section sizes and spans increased so did the concentrated load at the 1.5-inchwide load bearing plates. This increased the local bucking effects and initiated failure in the compression flange before yielding in the tension flange or lateral-torsional buckling of the whole member was experienced. However, local buckling of the top flange initiated a bending type of failure even in the short span specimens. As expected, the h/t shear reductions for the individual members using the AISI Specification were offset by the composite effects of the back-to-back header assembly. This is evident in the data in Table 3 because the "load increase ratios" are largest for header assemblies with the largest h/t ratios (Tests C, E, and G).

As seen in Table 3, bending controlled the calculated design capacity in only the 2-2x4x43 (Test B) header assembly. This is due to the relatively long span with a low h/t ratio. This member also had the smallest h/t ratio and the smallest load increase ratio. As previously stated, failure was initiated by local buckling of the compression flange due deformation of the flange under concentrated load applied through the 1.5-inch-wide bearing plates. Therefore, it is unclear that adding traditional web stiffeners at the load points (with typical construction practices and

tolerances) would have dramatically improved performance such that the failure mode may have been yielding or lateral torsional buckling. Greater distribution of the load (using more than a two-point load) may have improved tested capacities. The 1.22 load increase ratio for the 2x4x43 (Test B) test specimen could be due to the fact that the calculated values did not include composite action from the tracks attached to the top and bottom of the header flanges. Therefore, one could conclude that the bending capacity is predicted rather well and that a small load increase factor (e.g., 20 percent) may be appropriate when the headers are attached to track members on the top and bottom flange. However, since the failure mode was local buckling initiated, it is difficult to substantiate such a system effect from these tests. In summary, the local buckling failure mode indicates that other factors play a much less significant role than steel thickness and local buckling due to deformation in the flange caused by concentrated loads for the condition of these tests.

CONCLUSIONS

The following conclusions are supported by the findings of this work:

- The practice of designing built-up header members by doubling single-member capacities results in very conservative spans.
- The largest increases in tested capacity relative to predicted capacity were realized for the back-to-back header specimens made from individual members with high h/t ratios and shorter spans (i.e. the single member design capacity was controlled by shear).
- The smallest increases in tested capacity relative to predicted capacity were realized for the back-to-back headers with longer spans and smaller h/t ratios (i.e., the single member design capacity was controlled by bending).
- The consistent failure mode in <u>all</u> tests was related to local buckling of the top (compression flange) at the concentrated load points.
- Greater tested capacities would undoubtedly have been achieved with a more uniform load distribution rather than the load created by the 2-point load apparatus, particularly for longer header spans that are intended for use in a 24 inch on-center framing system.

RECOMMENDATIONS

Although the results are very promising, more testing and analysis of the data is needed to develop an accurate design procedure for the built-up header assemblies. The following recommendations are suggested for future tests and analysis:

- use larger load bearing plates to minimize local buckling effects and isolate other failure modes (i.e. bending, shear, etc.), although this would not be representative of a repetitive member framing system;
- brace the compression flange every 24 inches if the span is over 48 inches;
- a wider range of span lengths should be tested for each specimen size;
- test each specimen for shear and bending failures separately;
- develop improved h/t relationships for built-up sections;



- re-evaluate web crippling and h/t limitations in the AISI Design Specifications; and
- investigate a design approach using a true back-to-back section model in lieu of doubling the capacity of a single member model.

These additional tests would create a comprehensive data set from which improved design rules could be created and developed for use in the AISI Design Specification and future editions of the Prescriptive Method for back-to-back headers constructed from C-sections. The potential design and construction cost savings are clearly evidenced in the findings of this study.

REFERENCES

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Header Test Results and Comparisons

Deflection L/

Г	co.	D	-	-				—	-	—	F								Г		100					.50			100		Г			rcle	П
	Controls	Calculated	Design	SHEAR	SHEAR	SHEAR	SHEAR	MOMEN	MOMENT	MOMEN	MOMEN	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR	SHEAR								
Calculated Ratio Tested	Q	Calculated	Ultimate Load	1.52	1.40	1.43	1.45	1.21	1.23	1.23	1.22	4.53	3.92	4.45	4.30	1.48	1.68	1.63	1.60	2.57	2.61	3.33	2.83	1.43	1.44	1.37	1.41	3.60	3.97	3.27	3.61	1.83	1.97	1.72	1.84
Calculated	Ultimate	Load	(Ibs)	2,687	2,687	2,687	2,687	3,124	3,124	3,124	3,124	1,707	1,707	1,707	1,707	3,846	3,846	3,846	3,846	3,214	3,214	3,214	3,214	6,139	6,139	6,139	6,139	3,246	3,246	3,246	3,246	11,654	11,654	11,654	11,654
Calculated	Ultimate	Shear	(sql)	1,343	1,343	1,343	1,343	2,986	2,986	2,986	2,986	854	854	854	854	1,923	1,923	1,923	1,923	1,607	1,607	1,607	1,607	3,070	3,070	3,070	3,070	1,623	1,623	1,623	1,623	5,827	5,827	5,827	5,827
Calculated	Ultimate	Moment	(ft-lb)	1,917	1,917	1,917	1,917	2,473	2,473	2,473	2,473	4,964	4,964	4,964	4,964	6,772	6,772	6,772	6,772	8,552	8,552	8,552	8,552	12,369	12,369	12,369	12,369	12,972	12,972	12,972	12,972	22,776	22,776	22,776	22,776
Percent	Elongation	(2" gauge	length)	23%	27%	%07	23%	27%	%97	27%	27%		28%	23%	25%	27%	798	24%	792			792	798	14%	7%	13%	11%	10%	15%	13%	13%	18%	19%	19%	19%
Measured	Ultimate	Stress	Fu² (ksi)	55.96	58.26	58.28	57.50	53.18	53.46	53.47	53.37	56.53	56.14	56.13	56.27	55.95	55.68	56.27	55.97	54.56	54.57	54.87	54.67	70.90	70.90	68.84	70.21	66.48	66.48	66.17	66.37	70.77	68.97	69.25	99.69
	Measured	Yield Stress	Fy ² (ksi)	48.93	50.16	49.39	49.49	47.86	47.47	49.07	48.13	53.68	52.87	53.30	53.28	51.95	51.04	51.58	51.52	49.90	49.25	50.33	49.83	68.10	68.10	86.98	67.72	66.48	66.48	66.17	66.37	65.13	62.88	65.17	64.39
Measured	Base Metal	Thickness ²	(in)	0.0327	0.0321	0.0326	0.0325	0.0433	0.0434	0.0432	0.0433	0.0323	0.0328	0.0325	0.0326	0.0424	0.0431	0.0427	0.0427	0.0428	0.0433	0.0429	0.0430	0.0536	0.0536	0.0514	0.0529	0.0419	0.0425	0.0427	0.0424	0.0684	0.0691	0.0688	0.0688
	ž	ratio		115	117	115	115	98	86	98	98	240	236	238	238	182	179	181	181	227	225	227	226	181	181	189	183	280	276	275	277	170	168	168	169
Load at	Allowable	Deflection	(lbs)	×	×	×	×	×	×	2,595	2,595	7,430	5,722	7,598	6,917	4,552	5,030	5,750	5,111	×	×	8,239	8,239	7,926	7,630	6,345	7,300	×	×	8,096	960'8	17,710	16,440	19,340	17,830
	Ultimate Standard Allowable	Devlation Deflection Deflection	(In)	0.1958	0.1958	0.1958	0.1958	0.2375	0.2375	0.2375	0.2375	0.1500	0.1500	0.1500	0.1500	0.3333	0.3333	0.3333	0.3333	0.2792	0.2792	0.2792	0.2792	0.4333	0.4333	0.4333	0.4333	0.2375	0.2375	0.2375	0.2375	0.4333	0.4333	0.4333	0.4333
	Standard	Devlation	(sql)				157.41				41.62				560.23				408.19				1386.06				243.79				1150.93				1440.36
Test	Ultimate	Load	(lbs)	4,076	3,774	3,848	3,899	3,774	3,848	3,844	3,822	7,726	869'9	7,598	7,341	5,682	6,464	6,276	6,141	8,246	8,378	10,710	9,111	8,766	8,856	8,396	8,673	11,670	12,900	10,600	11,723	21,270	22,920	20,050	21,413
	Measured	Dimensions		47 1-5/8"x3-1/2"	47 1-5/8"x3-1/2"	47 1-5/8"x3-1/2"	47 1-5/8"x3-1/2"	1-5/8"x3-1/2"	1-5/8"x3-1/2"	1-5/8"x3-1/2"	1-5/8"x3-1/2"	1-5/8"x8"	1-5/8"x8"	1-5/8"x8"	,1-5/8"x8"	1-5/8"x8"	1-5/8"x8"	1-5/8"x8"	1-5/8"x8"	1-5/8"x10"	1-5/8"x10"	1-5/8"x10"	1-5/8"x10"	1-5/8'x10"	1-5/8"x10"	1-5/8"x10"	1-5/8"x10"	1-5/8"x12"	1-5/8"x12"	1-5/8"x12"	1-5/8'x12"	1-5/8"x12"	1-5/8'x12"	1-5/8'x12"	1-5/8'x12"
		Span	(inches)	47	47	47	47	22	22	22	22	36	36	36	36	80	80	80	80	29	29	29	29	104	104	104	104	22	22	22	25	104	104	104	104
			Test (2x4x33 test 1a	2x4x33 test 1b	2x4x33 test 1c	Test 1 average	2x4x43 test 2a	2x4x43 test 2b	2x4x43 test 2c	Test 2 average	2x8x33 test 8a	2x8x33 test 8b	2x8x33 test 8c	Test 8 average	2x8x43 test 7a	2x8x43 test 7b	2x8x43 test 7c	Test 7 average	2x10x43 test 4a	2x10x43 test 4b	2x10x43 test 4c	Test 4 average	2x10x54 test 6a	2x10x54 test 6b	2x10x54 test 6c	Test 6 average	2x12x43 test 3a	2x12x43 test 3b	2x12x43 test 3c	Test 3 average	2x12x68 test 5a	2x12x68 test 5b	2x12x68 test 5c	Test 5 average

Notes:

The failure mode was local buckling of the compression flange at one of the two concentrated loads for all tests.
 Measured yield stress, ultimate stress, percent elongation and thickness represent the average of three coupon samples taken from the lot of the specified material.
 The 2x12x43 section ht ratio exceeds 260. Therefore, a thickness of 0.0451" was used for calculation purposes due to limitations in the analysis software.
 All return lips were 1/2 inch. Dimensions were verified to be within a tolerance of +/- 1/16 inch.

Given:

Section F	roperties are taken from	Section Properti	ies table		
Sx =	0.566 in^3		29,500,000 psi	Sf=	0.811725 in^3
Fy =	53,300 psi	pucnchout=	1.5 in	rx =	2.9415 in
h =	8 in	R =	0.09375 in	ry =	0.5499 in
t =	0.0326 in	K =	. 1	ro =	3.141 in
Fyc =	53,300 psi	L=	12 in	J =	0.000138 in^4
Pi =	3.14159	A =	0.3889 in^2	G =	11,300,000 psi
Cw =	1.5257 in^6	lxx(gross)=	3.2469 in^4	Ycg(gross)=	4
IXX =	3.2469 in^4	lyy(gross)=	0.1176 in^4		

C3 Flexural Members

Find Allowable Bending Moment (Ma)

Ma=Mn/Saftey Saftey= 1.67 Ma=Mn/1.67

Mn is the smalest of C3.1.1, C3.1.2 and C3.1.3

C3.1.1 Nominal Section Strength Based on Iniation of yielding

Mn=SeFy

Mn= 30,167.80 in-lb

18064.55 1505.379

C3.1.2 Lateral Buckling Strength

Mn=ScMc/Sf

For Me>.5My

Mc=My(1-My/4Me)

My=SfFy

My= 43,264.94 in-lb

Me= CbroA(OeyOt)^.5

Cb= 1 (M for unbraced length is larger than M @ ends)

Oey=Pi^2E/(KxLx/rx)^2

Oey= 611,400.76 psi

Ot=(1/Aro^2)(GJ+Pi^2ECw/(KtLt)^2

Ot=

804,401.80 psi

Me=

856,653.87 in-lb

Mc1=My(1-My/4Me)

Mc1=

42,718.67 in-lb

For Me<=.5My

Mc2=Me

Mc2=

856,653.87 in-lb

Mc=

42,718.67 in-lb

Mn=

29,786.90 in-lb

C3.1.3 Beams having one flang through-fastened to Deck or Sheathing

Does not apply

Mn=

29,786.90

Ma=

17,836.47 in-ib

Ma=

1,486.37 ft-lb

C3.2 Strength for Shear (unpunched)

Compute the depth of the flat portion of the web(H)

H=h-2(R+t)=

7.7473

H/t= 237.64724

Calculate 1.38(Ekv/Fy)^.5

5.34

 $1.38(Ekv/Fy)^{.5} =$

75.02346

If H/t<=

75.023465 lb

 $Va1 = .38t^2(kvFyE)^5 <= .4Fyht$

Va1 =

1,170.21 lb

If h/t >

75.023465

 $Va2 = .53Ekvt^3/H$

Va2 =

373.37 lb

Va=

373.4 lb

ICBO method for calculating Shear (punched)

Caculate reduction factor

qs = 1-1.1(a/d)

qs =

0.79375

Va =

296.4 lb

Header Comparisons:

Span =

36

inches

Vu2 = 2*Va*1.44

Vu2 =

853.53 lb

Ma2 = 2*Ma*1.67

Ma2 =

4964.48 ft-lb



Puv = 2*Vu2

Puv =

1707.06 lb

Pum = 6*Mu2/L

Pum =

9928.97 lb

Pu =

1,707.06 lb



1986 AISI Specification w/1989 Addenda DATE: 7/7/97 NAHB Research Center

SECTION DESIGNATION:

INPUT PROPERTIES:											
Web Height = Top Flange =	8.000 in 1.625 in 1.625 in 0.500 in	Steel Thickness = Inside Corner Radius = Yield Stress, Fy =	0.0326 0.0938 53.3	in							
Punchout Width =	1.500 in	Punchout Length =	4.000	in							
OUTPUT PROPERTIES:											
Effective Section Proper Neutral Axis from To Moment of Inertia fo Section Modulus (S: Allowable Bending M	op Fiber (Ycg) or Deflection (Ixx) xx)		4.7091 3.2469 0.5660 1505.45	in^4 in^ 3							
Gross Section Properties Neutral Axis from To Moment of Inertia (I) Cross Sectional Are Radius of Gyration (op Fiber (Ycg) xx) a (A)	Strong Axis	4.0000 3.3652 0.3889 2.9415	in^4 in^2							
Gross Neutral Axis (Gross Moment of In Radius of Gyration (Effective Section Mo Effective Neutral Ax Allowable Moment (I	(Xcg) From Web F ertia (lyy) (Ry) odulus (Syy) is (Xcg) from Web		0.3454 0.1176 0.5499 0.0734 0.7789 195.20	in^4 in in^3 in							
Other Section Property D Net Area at Punchor Member Weight per Allowable Shear For Allowable Shear For Pao for use in Intera	uts Foot of Length rce In Web (Unpur rce In Web (Punch	ned)	0.3400 1.2590 373.37 296.36 3791	lb/ft lb lb							
Torsional Properties Dist. from Shear Cer St. Venant torsion C Warping Constant (C Radii of Gyration (Re Torsional Flexural C	onstant (J x 1000) Cw) o)		-0.9546 0.1378 1.5257 3.1410 0.9076	in^4 in^6							

^{****} WEB DEPTH-TO-THICKNESS RATIO EXCEEDS 200 ****



METRIC CONVERSIONS

1 mil = 1/1000 inch

1 kip = 1000 lbs = 4.448 kN

1 inch = 1000 mils = 25.40 mm

AISI

RG-9719 0897-2.5M-RI

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